

The bittersweet conclusion time in space

May 21st, 2018 – Dec 24th, 2021



Once the the altitude dropped below 320 km things deteriorated faster than expected Same late mission experime

Some late mission experiments could not be completed

After solving so many challenges,

we were finally "harvesting" Come on...wasn't 2020

Come on...wasn't 2020 already bad enough?

There isn't really much we could have done about it

The job was done, and then some

Many successors of RainCube are

already growing up

Yep, it was bad, and all this is nothing, considering.

RainCube in a nutshell

Technological goal (of the InVEST'15 tech demo):

Demonstrate the first active remote sensing instrument in a CubeSat, via a Ka-band precipitation radar

Scientific goal (of a constellation or train):

To provide global observations of the temporal evolution of vertical structure and thermodynamic processes of storms.

Success criteria & relevance of the timeline:

Detect precipitation & capture vertical structure of storms.

Do so in a timely fashion to inform the

Cloud, Convection and Precipitation studies prompted

~2.0U

by the 2017 Earth Science Decadal Survey.

miniKaAR-C (radar electronics)

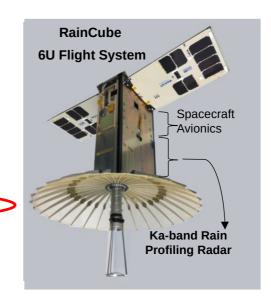
Reduced size, weight, and power by offset IQ with pulse compression produlation technique

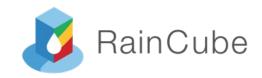
KaRPDA (antenna)

Half meter Ka-band lightweight deployable

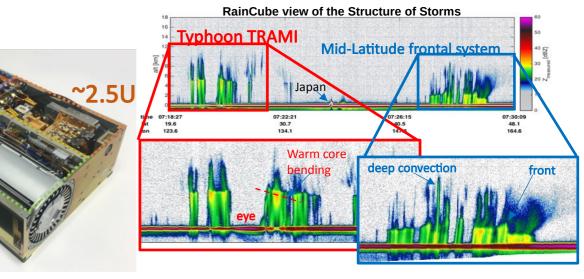
Tyvak Bus:

Compact highly integrated bus providing 35W of power to the payload.







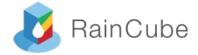


Peral E., S. Tanelli, S. Statham, S. Joshi, T. Imken, D. Price, J. Sauder, N. Chahat, A. Williams, "RainCube: the first ever radar measurements from a CubeSat in space," J. Appl. Remote Sens. 13 (3), 032504 (2019), doi: 10.1117/1.JRS.13.032504.

Radar Electronics & Antenna Reqs. (4U)

Req't Name	Requirement	Measured		
Sensitivity @400km	20dBZ	11.0dBZ		
Horizontal resolution @400km	10km	7.9 km		
Nadir Data Window	0-18 km	-3 to 20 km		
Vertical resolution	250m	250m		
Downlink data rate (in transmit)	50 kbps	49.57 kbps		
Payload power consumption (AntDeployment/STDBY/RXOnly/TXScience)	10/8/15/35 W	5/3/10/22 W		
Mass	6 kg	5.5 kg		
Range sidelobe suppression	>60dB @ 5km	>65dB @ 1km		
Transmit power & Transmit loss	10W / 1.1dB	>39dBm		
Antenna gain	42 dB	42.6 dB		
Antenna beamwidth	1.2 deg	1.13 deg		

System Architecture



Deployable Deployable Solar Arrays
UHF Antenna

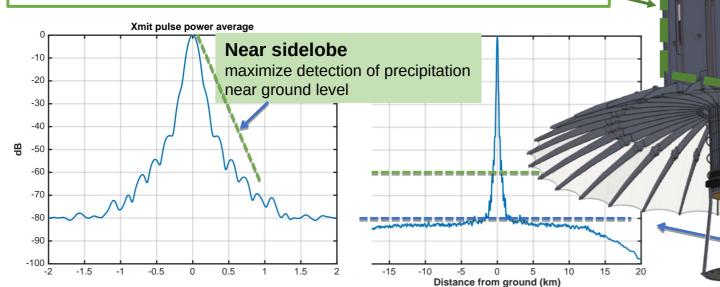
Bus Reqs. (2U)

- Provide 35 W for payload power in transmit mode
 - Maintain a 25% radar duty cycle
- 12.1 Gb/week of payload data
- Maintain payload temperatures (-5C to +50C operational)
- GPS provides on-board altitude to radar

S-Band Patch Antenna & Transmitter

Deployable
0.5 m KaRPDA
Radar Antenna

Far sidelobes below -60dB for clutter-free detection in all conditions throughout the troposphere



How small is RainCube. . .

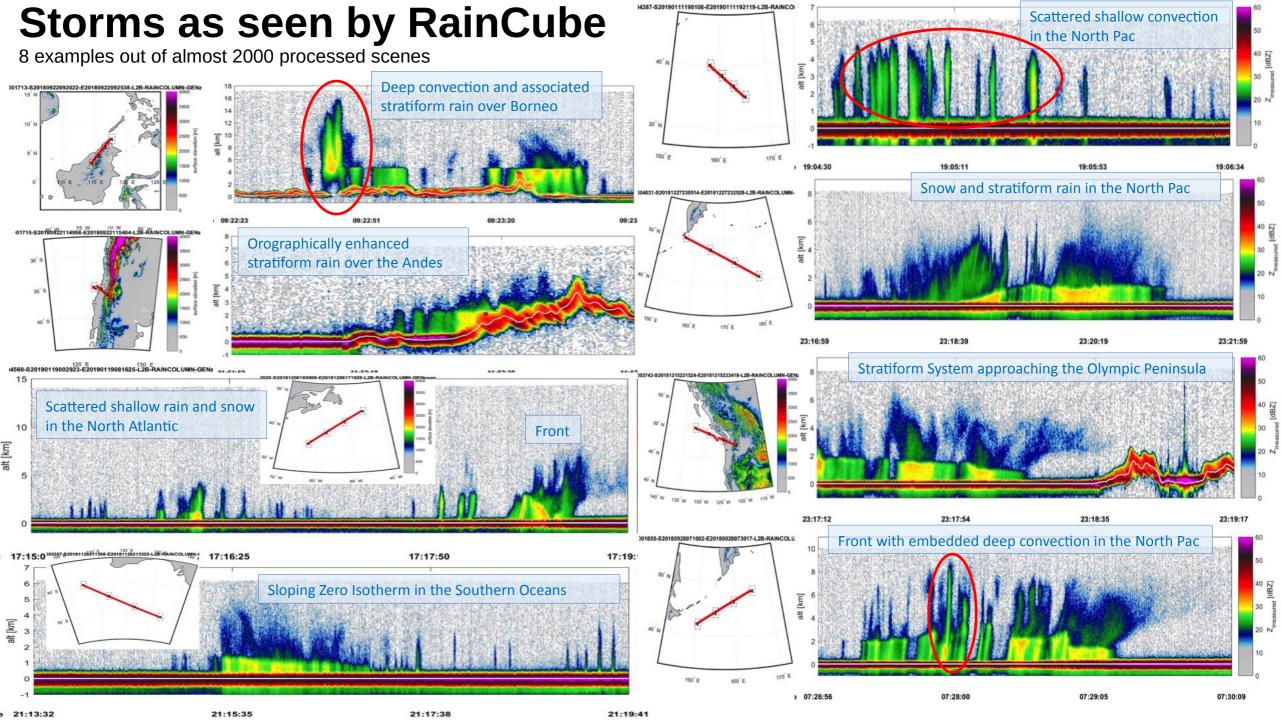


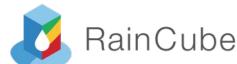


Radar

RainCube Ka-band Precipitation

	CPR	KaPR	RainCube	
Mass [Kg]	260	336	7	
Power [W]	300	344	22	
Volume [U]	4,356	1,210	4	
Class	С	С	Tech demo	
Frequency	W-band	Ka-band	Ka-band	
Scanning	No	Yes	No	
Sensitivity	-30 dBZ	+17 dBZ	+12 dBZ	





RainCube What an in-orbit technology demonstration could do in 2 years

- RainCube operated in space between Aug 2018 and December 2020
 - It re-entered the atmosphere on Dec 24th, 2020 with the Radar still fully functional
- Assumed additional scope as soon as the prime objective was completed
 - Mission success could be claimed after the first storm detection.
 - Past that milestone, risks associated with additional objectives could be evaluated
 - Operating the mostly-COTS radar over the SAA
 - Scheduling data takes to target specific storms based on forecast
 - Acquiring data in proximity of GPM for cross comparison and validation
 - Devising a new ADCS algorithm solution to maintain fine pointing with only 2 remaining reaction wheels.
 - Adopting the Amazon Web Services for global downlink capability
 - Designing and implementing new waveforms to validate end-to-end performance models
- Operated with a very small team to achieve the desired goals
 - Many early career team members could benefit of invaluable experience somewhere between "technology development" and "flight project", and with a rewarding short turnaround.
- Enabled and sometimes forced creative solutions throughout its lifecycle.
- Demonstrated a number of technology solutions which in turn enabled a number mission and instrument concepts where either all or only one of them are necessary.
 - Inspired a number of groups, worldwide, to pursue similar challenges and bring it to the next level.
- Provide science data (Available on: https://tcis.jpl.nasa.gov/data/raincube/)

Main outcomes

Demonstration of feasibility of mission concepts involving multiple radar platforms

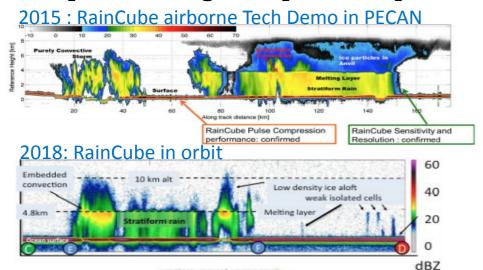
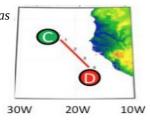


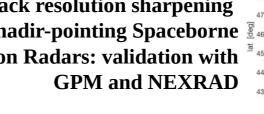
Fig. 2. (a) MiniKaAR-C data from the airborne campaign as a part of the PECAN campaign (June 28–July 15, 2015) showing the profile of radar reflectivity Z. The reflectivity color scale is in dBZ. The expected on orbit sensitivity of Raincube of \sim 10 dBZ is emphasized with the color scale chosen. (b) MiniKaAR-C reflectivity profiles of rain falling from the nascent hurricane florence.

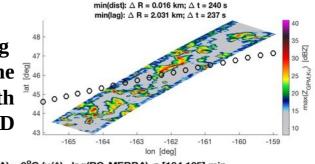


Stephens G.L., S.C. van den Heever, Z. S Haddad, D J Posselt, R L Storer, L D Grant, O O Sy, T. Narayana Rao, S. Kumar, S. Tanelli and E. Peral, "A distributed small satellite approach for measuring convective transports in the Earth's atmosphere", IEEE TGRS, Print ISSN: 0196-2892; Online ISSN: 1558-0644; doi: 10.1109/TGRS.2019.2918090

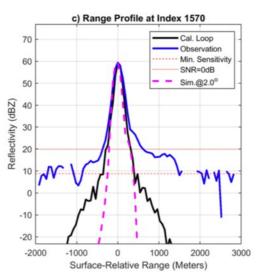
RainCube

Along-track resolution sharpening for nadir-pointing Spaceborne 34 **Precipitation Radars: validation with**

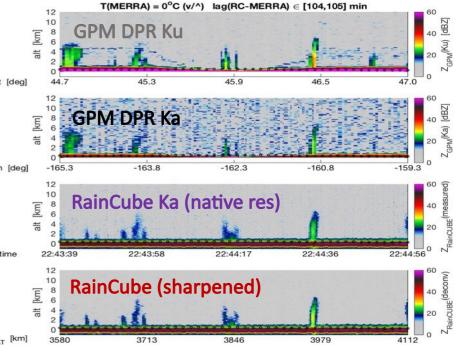




Validation of accurate modeling of pulse compression performance for Precipitation Radars from Low Earth Orbit lat [deg]

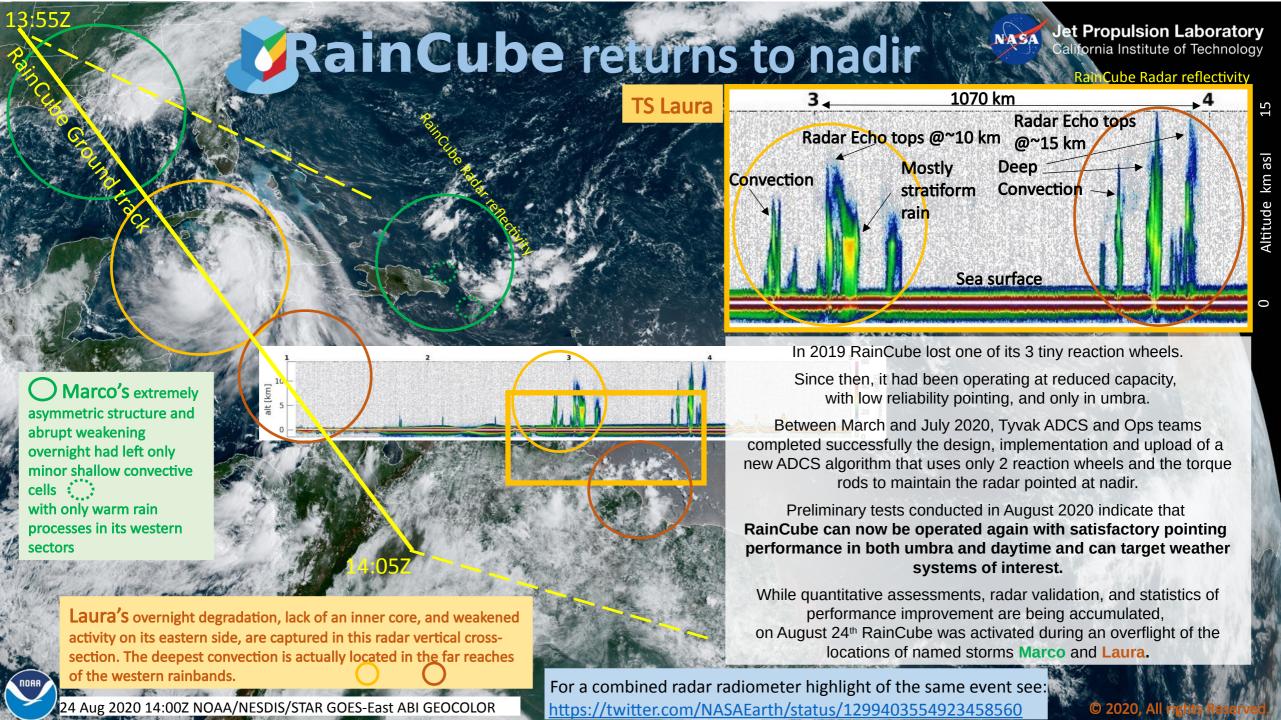


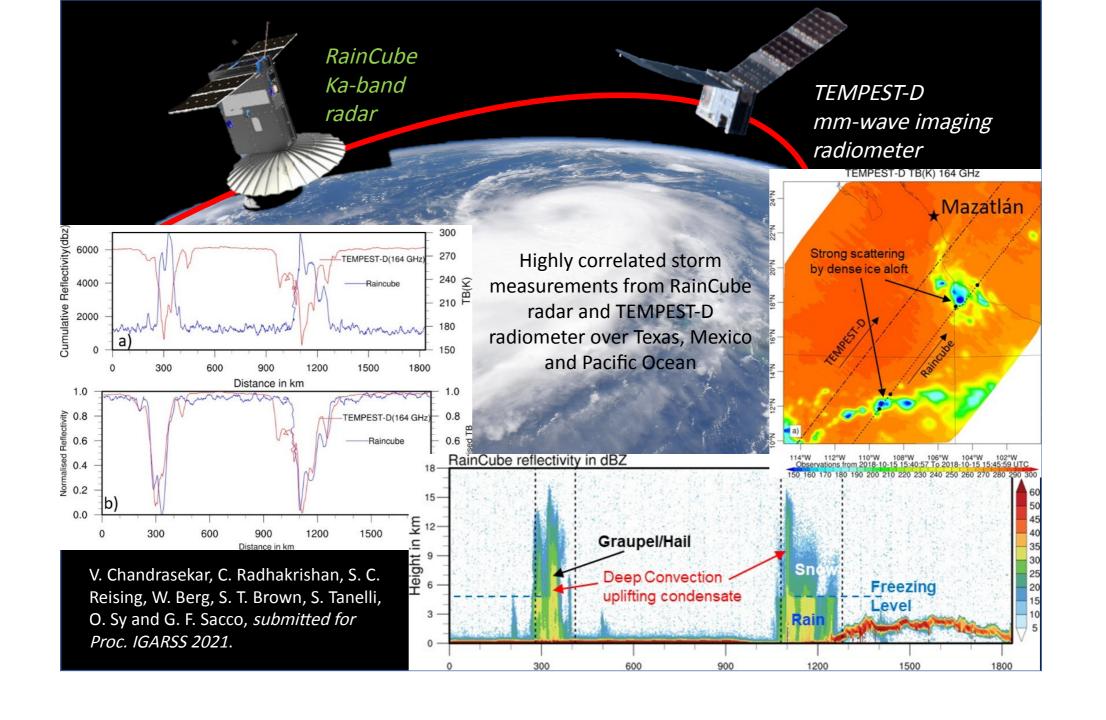
Beauchamp R.M., S. Tanelli and O.O. Sy, 2020: Observations and Design Considerations for Spaceborne Pulse Compression Weather Radar, in IEEE Transactions on Geoscience and Remote Sensing, vol. 59, no. 6, pp. 4535-4546, June 2021, doi: 10.1109/TGRS.2020.3013164.



O.O. Sy, S.Tanelli, G.S. Sacco, E.Peral, 2021: RainCube: first Cubesat demonstrator of deconvolution for spaceborne cloud and precipitation radar measurements, in IEEE Transactions on Geoscience and Remote Sensing, doi: 10.1109/TGRS.2021.3073990.

Tanelli S., E. Peral, O. O. Sy, G. F. Sacco, Z. S. Haddad, S. L. Durden, S. Joshi, "RainCube: How can a CubeSat radar see the structure of a storm?." Proc. SPIE 11131, CubeSats and SmallSats for Remote Sensing III, 1113106 (30 August 2019); https://doi.org/10.1117/12.2531150



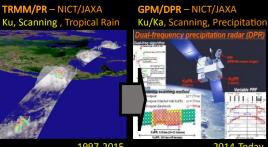


Spaceborne "Atmospheric Radar" landscape (2029)

* Pre-Decisional Information – For Planning and Discussion Purposes Only

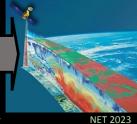
= Notional accommodation concept among several possible

The 4 "predecessor" Spaceborne Cloud & Precipitation Radars



1997-2015
t/CPR
A/CSA
BZ , Clouds

2014-Tod
EarthCARE/CPR
NICT/JAXA
W, Doppler, Clouds



IIP 2019: CloudCube
Ka/W/G, Scanning Ka, DPCA Doppler,
SmallSat

Solar panels
Solid 1.6 m Ka/W-band antenna

1 Ka/W-band centered feed horn

* ACCP (AtmOS) Candidate #1
Ku/W, Narrow Scanning Ku,
DPCA Doppler Ku

#

Science Models

Operational

RainCube

JPL/NASA InVEST Tech Demo Ka-band, Precipitation, 6U CubeSat

2018 – 2021

NOAA Architecture Studies

International developments in EU and Asia

Commercial endeavors

IIP 2019: VIPR (in-cloud Vapor profiling) & PBL radar *

Strengths:

- unprecedented view of the vertical structure of weather systems
- Improved quantitative precipitation estimates on a global scale
- Synergy with other instruments (radiometers, lidars, etc.)
- As ensemble: sufficient dynamic range to cover from nonprecipitating clouds to severe storms.

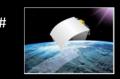
Weaknesses:

- Limited temporal coverage (singles in LEO)
- Limited spatial coverage (narrow or no swath, singles

ES DS 2007: ACE Mission Concept Radars *
and their evolution towards a wholistic
cloud/precipitation mission concept
Ku, Ka, W, Doppler, Scanning

* ACCP (AtmOS) Candidate #2
Ka /W, Narrow Scanning Ka,
DPCA Doppler Ka & W

IIP 2016: MASTR Ku/Ka/W, Scanning, Doppler, SmallSat



RainCube
Ka-band / 0.5 m

Ka-band
1.6 m Antenna
SmallSat

Ka-/W-band
Dual
2.1 m Antenna

RainCube's legacy

Pre-Decisional Information -- For Planning and Discussion Purposes Only

CloudCube

Ku-/W-band Dual 2.1 m Antenna ("SmallSat") Ka-/W-/G-band 1-2 m Antenna (SmallSat)

> # of sats 11-100

Operational monitoring (Combined w. Radiometers)

Diurnal Cycle sampling

Inspired:

NOAA Architecture

+ International Efforts
+ Commercial Enterprises

Operational model validation)

6-10

EV-class concept

Convective Mass Flux in

Tropical Storms

Tech Demo 2018-2020 ACCP Candidate Radar
For polar component
Broad cloud/precipitation
dynamics

ACCP Candidate Radar For inclined component Deep Tropical convection dynamics

Next gen

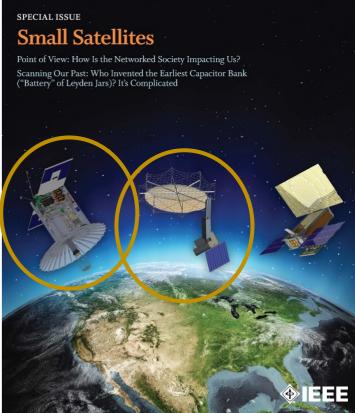
Cloud microphysics

and dynamics

2-5

RainCube has opened up the spectrum of possible cloud and precipitation radar solutions to adapt to a vast array of mission concepts





Constellation of RainCube's "as is"

RainCube: What's next?

Analyzing the current dataset to demonstrate the potential and the limitations of the current system in addressing science questions

Constellation with improved antennas & electronics

- To address a larger set of science questions
- Development of technologies and of mission concepts is ongoing
- Extension to W and G-band for cloud & precip.
- **DPCA** for Doppler, Larger Size for improved resolution and sensitivity, multi-feed for

RainCube 1.0 m DPCA in 24U RainCube 1.0 m





- Planetary applications

in 12 U



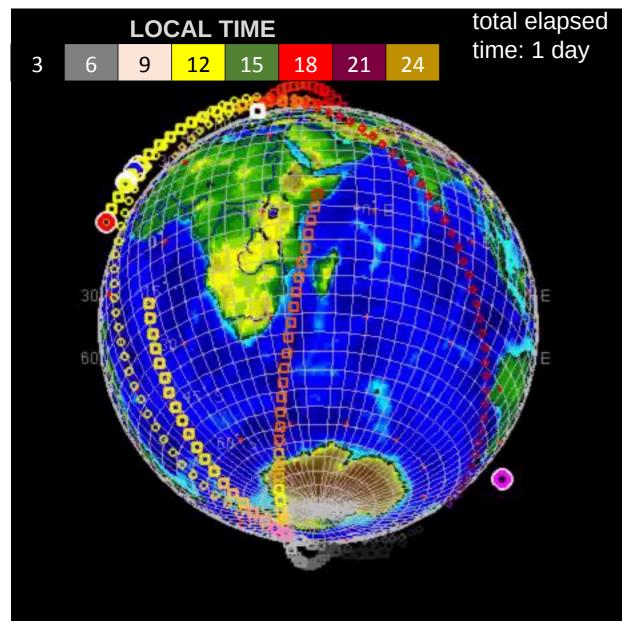




Ka-band ESTO InVEST and ACT programs

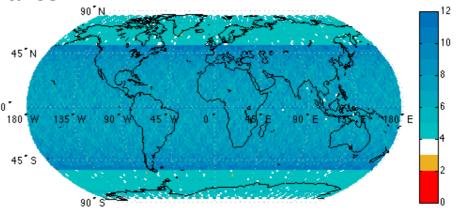
	6U	12 U	50 kg
Antenna size [m]	0.5	1.0	2.0
Sensitivity [dBZ]	15	5-10	0-5
Hor Resolution [km]	8	4	2
Range Res [m]	250		
Beams	1	1-3	1-5
RF Power [W]	10	10-20	10-40

RainCube InVEST'15 "as is": strength in numbers



Global sampling of the **diurnal cycle** of precipitation can be conducted effectively with a small number of simple radars.

The notional mission concept adopted in this particular simulation involves 3 distinct orbital planes.



Number of different hours of day in local time that a given 2° x 2° box was visited within a **16 day** period.

Almost everywhere in the globe, each box was visited at least at four different hours of the day.

This class of radar enables not only new research but can have an **application**-oriented role for operational weather agencies, as well as commercial sector enterprises.

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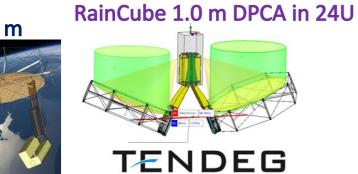
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RainCube 1.0 m

in 12 U









- Constellation with other Radars and Radiometers:
- A study team in the Earth Science Decadal Survey 2017 is considering RainCube-like constellations for measurements of convection and precipitation
- Higher frequency versions of RainCube for cloud and water vapor observations
- Planetary applications
 - An evolution of this instrument could support altimetry and cloud and precipitation on planetary targets

Ka-band ESTO InVEST and ACT programs

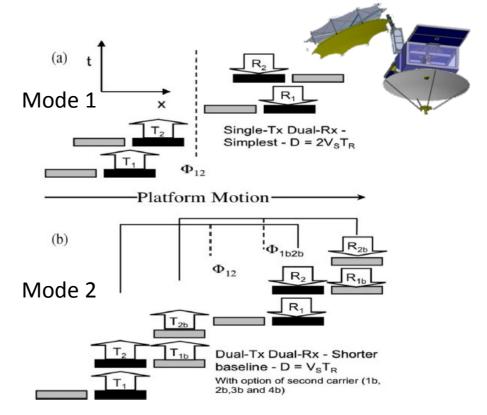
Ita balla Esto investanta Acti programs				
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Displaced Phase Center Antenna for high accuracy Cloud and Precipitation Doppler estimates

One upgrade in CloudCube wrt RainCube is that it can Tx and Rx alternatively from a Fwd and Aft antenna to adopt the **DPCA technique** [1,2].

DPCA effectively cancels (or reduces by one or more orders of magnitude) the effective platform velocity contribution to signal broadening:

- Reduction of spectral broadening
 - improvement in mean Doppler estimation precision
 - Improvement in estimation of the target's natural Doppler spectral width (i.e., turbulence, shear, hydrometeor size spread)
- Reduction in Non-Uniform Beam Filling biases [3-8]
 - improvement in mean Doppler estimation accuracy



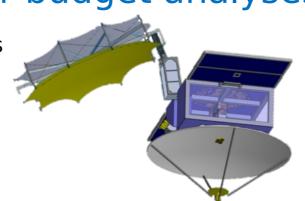
Tanelli, S.,

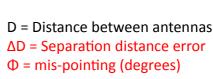
- . **Durden, S. L.**, **Siqueira, P. R.,** & Tanelli, S. (2007). On the use of multiantenna radars for spaceborne doppler precipitation measurements. IEEE Geoscience and Remote Sensing Letters, 4(1), 181–183.
- Tanelli, S., Durden, S. L., & Johnson, M. P. (2016). Airborne Demonstration of DPCA for Velocity Measurements of Distributed Targets. IEEE Geoscience and Remote Sensing Letters, 13(10), 1415–1419. [4] Im, E., Durden, S. L., Facheris, L., & Giuli, D. (2002). The effects of nonuniform beam filling on vertical rainfall velocity. Journal of Atmospheric and Oceanic Technology, 19(7), 1019–1034
- Tanelli, S., Im, E., Kobayashi, S., Mascelloni, R., & Facheris, L. (2005). Spaceborne Doppler Radar Measurements of Rainfall: Correction of Errors Induced by Pointing Uncertainties. Journal of Atmospheric and Oceanic Technology, 22(November), 1676-1690.
- 4. Schutgens, N. A. J., 2008: Simulated Doppler Radar Observations of Inhomogeneous Clouds: Application to the EarthCARE Space Mission. J. Atmos. Oceanic Technol., 25,26–42, doi:10.1175/2007JTECHA956.1.
- 5. Battaglia, A; Tanelli, S., "DOMUS: DOppler MUltiple-Scattering Simulator", IEEE Trans. Geoscience and Remote Sensing, vol.49, no.1, pp.442,450, Jan. 2011
- Kollias, P., S. Tanelli, A. Battaglia, A. Tatarevic, (2014): Evaluation of EarthCARE Cloud Profiling Radar Doppler Velocity Measurements in Particle Sedimentation Regimes. J. Atmos. Oceanic Technol., 31, 366–386. doi: http://dx.doi.org/10.1175/JTECH-D-11-00202.1
- 7. Sy, O.O.; Tanelli, S.; Takahashi, N.; Ohno, Y.; Horie, H.; Kollias, P., (2014). Simulation of EarthCARE Spaceborne Doppler Radar Products Using Ground-Based and Airborne Data: Effects of Aliasing and Nonuniform Beam-Filling. Geoscience and Remote Sensing, IEEE Transactions on, vol.52, no.2, pp.1463,1479, Feb. 2014. doi: 10.1109/TGRS.2013.2251639
- 8. Sy, O.O.; Tanelli, S.; Kollias, P.; Ohno, Y., 2014b, "Application of Matched Statistical Filters for EarthCARE Cloud Doppler Products", IEEE Trans. Geoscience and Remote Sensing, vol.52, no.11, pp.7297,7316, Nov. 2014

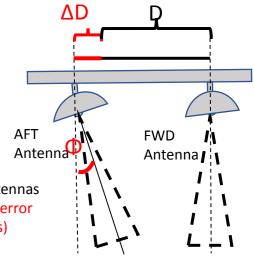
Displaced Phase Center Antenna error budget analyses

DPCA enables high quality Doppler measurements with 2 Antennas of less than L = 2.5 m diameter.

Compared to a single antenna solution with an antenna of say 5 m diameter:





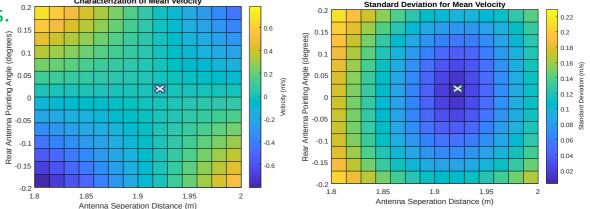


PROS:

- 1. Eliminates need to manufacture very large mm-wave reflectors.
- 2. Reduces significantly the 3rd antenna dimension (driven by f, via f/L).
- Eliminates NUBF biases.

CONS / MITIGATIONS:

- Requires PRI to match k*Vs*D
 - 1. Gentle degradation, adjustable PRI in orbit
- 2. Requires Antennas to be co-aligned
 - 1. Gentle degradation, only halving of standard antenna alignment tolerance
- 3. For specific combinations and sets of requirements, it can result in slow PRF which is then more prone to Doppler aliasing
 - 1. Coherency supports standard de-aliasing techniques (as in Ground Based Doppler Weather Radars)
 - 2. Selection between Mode 1 and Mode 2 enables multiple options with the same hardware.



Simulations by : **S. Graniello** (CSU/JPL, 2020)

Displaced Phase Center Antenna independent error analyses

Note the notional graphic rendering shown

here supports only Ku and Ka band DPCA.

Independent performance analyses are being carried out in the context of the ACCP Designated Observable Mission Architecture studies. This example is of of warm rain convective processes (Courtesy: **P. Kollias**, Stony Brook U.).

W-band DPCA is currently analyzed only for 2 convective processes (Courtesy: **P. Kollias**, Stony Brook U.). non-deployable antenna configurations. Ka-band measured, quality-controlled Doppler velocity W-band measured, quality-controlled Doppler velocity Height 60 65 70 30 35 45 60 65 70 Along track [km] Along track [km] Ka-band model, 'truth' Doppler Velocity at the radar resolution W-band model, 'truth' Doppler Velocity at the radar resolution 10 Height [km] 65 30 60 70 35 50 60 65 70 Along track [km] Along track [km]

Simulations confirm absence of NUBF-induced biases which would be otherwise affecting the measurement in a single-antenna configuration (see [3-8])

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RainCube 1.0 m DPCA in 24U







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in 12 U

- An evolution of this instrument could support altimetry and cloud and precipitation on planetary targets

Ka-band ESTO InVEST and ACT programs 6U 12 U 50 kg Antenna size [m] 0.5 1.0 2.0 Sensitivity [dBZ] 15 5-10 0-5 Hor Resolution [km] 8 4 2 Range Res [m] 250 Beams 1 1-3 1-5 RF Power [W] 10 10-20 10-40

